

Successful SES/IAD/SESA submission

FSG Academy – Main Workshop for FSG 2024 on 21st of October 2023 at Schaeffler in Herzogenaurach

Targets of the SES

The structural design of your vehicle, specifically frame, must be documented:

- The SES is a tool to guide you to provide the <u>relevant & important information</u> of your vehicle
- The SES is also a tool for the competition/us reviewers to ensure that your vehicle is:
 - rules compliant, but a passed SES does not guarantee a rules compliant design!
 - the proposed structural design is safe
- It is **not** your design tool. But, you can access:
 - equations to prove equivalence (to an unalloyed carbon steel tubular space frame, per T3.2)
 - general guidance notes
 - feedback
- The same applies to the IAD

WHAT GIVES PEOPLE FEELINGS OF POWER





Review order

- SES/IAD checked by multiple events/multiple reviewers.
 - First submit, first reviewed!
- You may upload after moving to the FSG registration, waiting or withdrawn list.
 - Don't wait until 10 minutes before the deadline
 - 2023: upload 12 hours earlier → first review 3 months earlier compared to last-second upload!

SES / IAD Reviewers wanted:

What I see on the FSG website: for all teams registered (!) all uploads – support needed

Account Overview	combustion	CV Deadline Uploads Changelog	
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- Want to become a SES / IAD Reviewer
 - Alumni, not an active team member anymore for at least 1 year+ (meaning, 'skip' a season)
 - Can also be a professional, non-alumni (Ulf)



General rules changes specific for SES/IAD

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Impact structures (T1.1.12, T3.2.1, T3.15, EV4.4.2, EV5.5.2, CV1.3.2)

T3.15 Implementation of impact protection

- Simplified and generalized wording
- Components must now also be protected if exposed from the front
 - Applies (among others) especially to accumulators at the sides





T3.15.3

Energy absorption needs to exceed 65 J

iversity	Øo, tube	Øi, tube	Support Span	absorbed Energy
	25,4 [mm]	22.2 [mm]	400 [mm]	76,964 [J]
	21.3 [mm]	16.1 [mm]	400 [mm]	64.140 [J]
	25.4 [mm]	22.2 [mm]	400 [mm]	70,293 [J]
	30.0 [mm]	26.0 [mm]	400 [mm]	64,195 [J]
	25.0 [mm]	21.4 [mm]	400 [mm]	49,200 [J]
	25.5 [mm]	22.1 [mm]	400 [mm]	76,660 [J]
	25.0 [mm]	21.0 [mm]	435 [mm]	76,710 [J]
	25.0 [mm]	21.5 [mm]	400 [mm]	75.478 UI
	30.0 [mm]	27.0 [mm]	404 [mm]	87.541 [J]
	25.0 [mm]	21.4 [mm]	430 [mm]	103.288 [J]
	25.0 [mm]	21.0 [mm]	400 [mm]	130 390 UI
	25.4 [mm]	22.2 [mm]	455 [mm]	58,282 (J)
	25.0 [mm]	21.4 [mm]	400 [mm]	98.620 UI
	25.0 [mm]	21.4 [mm]	495 [mm]	82,530 [J]
	25.0 [mm]	21.0 [mm]	400 [mm]	53,200 [J]
	25.4 [mm]	22.2 [mm]	400 [mm]	61 779 [1]
	25.0 [mm]	21.0 [mm]	400 [mm]	73,986 [1]
	25.0 [mm]	21.0 [mm]	460 [mm]	33,400 [1]
	25.0 [mm]	21.0 [mm]	400 [mm]	117,850 [1]
	25.4 [mm]	22.4 [mm]	400 [mm]	117,000 [1]
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	25.0 [mm]	21.0 [mm]	400 [mm]	74 200 [1]
	27.0 [mm]	23.0 [mm]	400 [mm]	82 535 [1]
	25.0 [mm]	21.4 [mm]	430 [mm]	121 974 [1]
	25,0 [mm]	21.0 [mm]	400 [mm]	130 820 [1]
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	25/4 [mm]	25.0 [mm]	450 [mm]	133 612 [1]
	25,0 [mm]	21.4 [mm]	405 [mm]	45 500 10
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	27,0 [mm]	24.0 [mm]	400 [mm]	120 680 [1]
	25,0 [mm]	24,0 [mm]	A60 [mm]	77 767 [1]
	25,0 [mm]	21,0 [mm]	400 [mm]	77,707 [J]
	23,0 [mm]	180(mm)	400 [mm]	04 565 (1)
	25,0 [mm]	18,0 [mm]	400 [mm]	94,505 [1]
	25,0 [mm]	21,0 [mm]	(mm) 000	55,200 [J]
	25,0 [mm]	21,3 [mm] 22.2 [mm]	420 [mm]	60,550 [1]
	25,4 [mmh]	21.0 [mm]	420 [mm]	50,500 [J]
	25/0 [mm]	21,0 [mm]	484 [mm]	30,300 [1]
	25,0 [mm]	21,0 [mm]	400 [mm]	77,300 [J]
	25,0 [mm]	21,3 [mm]	400 [mm]	66,292 [1]
	25,0 [mm]	21,0 [mm]	400 [mm]	84,179 [J]
	25,4 [mm]	22,2 [mm]	400 [mm]	85,390 [J]
	25,0 [mm]	21,0 [mm]	400 [mm]	74,900 [J]
	25,4 [mm]	22,2 [mm]	[mm] 004	78,120 [J]
	25,0 [mm]	20,0 [mm]	400 (mm)	151,584 [J]
	25,0 [mm]	21,0 [mm]	400 (mm)	85,090 [J]
	25,0 [mm]	21,0 [mm]	418 [mm]	69,700 [1]
	25,4 [mm]	22,4 [mm]	400 (mm)	96,216 [J]
	25,0 [mm]	20,0 [mm]	400 [mm]	77,100 [J]
	25,4 [mm]	22,2 [mm]	400 [mm]	139,900 [J]
	25,0 [mm]	21,0 [mm]	400 [mm]	125,000 [J]
	25,4 [mm]	22,2 [mm]	400 [mm]	66,510 [J]
	25,4 [mm]	22,1 [mm]	400 [mm]	71,100 [J]

Car City





Asymmetrical lay-up

T 3.4.4 Asymmetrical lay-up restriction

- If an asymmetrical lay-up is used in the primary structure, the thinner skin must have a thickness of at least 40 % of the thicker skin
- Value of 40% chosen basis wrt. thicker lay-ups of for example 2.5-3.0mm, resulting in 1.0-1.2mm needed o other layer. Prevention of <0.5 mm lay-ups used.

Intent

Prevent very thin (=unsafe) layups in critical locations for real application.



Quasi-isotropic laminates

T 3.5.4 Non quasi-isotropic laminate definition

When a laminate is not quasi-isotropic, i.e. has equal strength and stiffness in the 0°, 90° and $\pm 45^{\circ}$ direction, the results from the 3 point bending test will be assigned to the 0 • lay-up direction. To show equivalence in the SES, the 0° lay-up direction must be used and oriented accordingly in the chassis.'





in the car.

Intent



Successful SES submission

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Read Me, Version History and Rules Clarifications

- Read through the revision notes
- Upload new (revised) version and include a description of what has been changed in <u>TAB ' Version History'</u>
- Include relevant rules questions in text form (no screenshot/image) in <u>TAB 'Rules Clarifications'</u>
- Feel free to contact <u>ses@formulastudent.de</u> if you have found issues and/or have suggestions



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Cover Sheet

- Print on large sheet for inspection to enable quicker review
- Cells O24-O49: please read the information given in cell O21
- Do not forget about the attachment checklist:
 - Include your receipts and datasheets etc.
- Reviewers can quickly see, what the overall <u>safety margins</u> are within your chassis and where those margins are lower (needs more attention)



Chassis Pics

- Main issues:
 - Not actually filled out the check boxes, nor the required information provided in the sheet
 - Material dimensions and angles of FH, MH, MHB not actually shown
 - Cut-outs larger than 60 mm² not shown
 - Positioning of TSAC/TS not clearly shown in all required views (ISO, side, top)
 - Values of 'YourMaterialX' inner/outer skin and core thickness not in accordance with the value entered in the 3-point-bending test and/or skin shear test
 - The colours given in the chart were not applied in the required different views of the chassis (please do not choose your own)
 - A different format for showing the laminated structures was used, please keep to the given format (!),
 BUT feedback welcome (see example next slide)

University NameCar No.(s) & Event(s) Please attach pictures of the frame and/or monocoque in the table below for review during the SES proc All tubes/lay-ups must be colour coded to show outer diameter and wall thickness. Three view d isometric views of the structure (CAD_FEA models_etc) are accentable. Note: Identical composite I	
University Name Car No.(s) & Event(s) Please attach pictures of the frame and/or monocoque in the table below for review during the SES proc All tubes/lay-ups must be colour coded to show outer diameter and wall thickness. Three view d isometric views of the structure (CAD_EEA models_etc) are accentable. Note: Identical composite I	
Please attach pictures of the frame and/or monocoque in the table below for review during the SES proc All tubes/lay-ups must be colour coded to show outer diameter and wall thickness. Three view d isometric views of the structure (CAD_FEA models, etc) are acceptable. Note: Identical composite I	
the identical colour code beyond borders of their specific lamiante structure! Maybe you need with two picture sets (Structures & your design)	ess. Irawings ar ayups nee to work
Images must include dimensions/labels indicating the following:	Compliane shown?
Angle of main and front hoops, including angle of main hoop below upper side impact tube.	No
Angle between main hoop bracing and main hoop	No
Distance from top of main hoop to main hoop brace attachment	No
Distance from top of front hoop to front hoop brace attachment	No
Outer diameter and wall thickness of all tubes / monocoque lay-up	No
Compliance to T 7.3.1 - Protection of lubrication systems	No
Teams entering cars with IC Powertrains must show the location of the fuel tank and complete filler neck inside rollover protection structure (CV2.2.2) in all images and highlight them in the colour red.	No
Teams entering cars with EV Powertrains must show the location of all HV components in these images and highlight them by colouring them orange.	No
Teams with breakthroughs/cutouts/holes in the laminated primary structure greater 60 mm ² must show their location in these images and highlight them by colouring them purple. These breakthroughs/cutouts/holes must also be shown in the SE3D-file.	No

This chart below is	s for laminated	I members of the p	rimary structure, A	CPS and TSF	PS only!		
Structure	Thickness outer skin [mm]	Juter skin ayup	0 ore m aterial	C ore thickness	nner skin ayup	Thickness nner skin [mm]	Proof of T3.4.3
MHBS	2,00			15,0		2,00	
FHB	2,00			18,0		2,00	
FBH	2,00			25,0		2,00	
FBHS	2,00			20,0		2,00	
SIS vertical	1,00			18,0		1,00	
SIS horizontal	1,00			10,0		1,00	
SHB	2,00			18,0		2,00	
YourMaterial7							
YourMaterial8							
YourMaterial9							
YourMaterial10							
YourMaterial11							
YourMaterial12							

Printed out version of SES available: if you have suggestions how to format this chart, feel free to comment!

Chassis Pics

- For a reviewer one of the most important TABs:
 - Quick overview
 - If well structured, can make the review much quicker!
 - Other TABs effectively a repetition/detailed info of what is presented here!
 - Skin thickness of >1mm in combination with certain prepregs/fibers gives reviewer assurance. Skin thickness <0.7mm will trigger reviewer to look more closely.

Plyb	book Composite 1 (SIS ver	tical)
Ply Nr.	Material	Direction
1	SIGRAPREG® C W200-TW2/2-E323/45%	45/-45 dea
2	SIGRAPREG@ C U255-0/NF-E310/30%	0 deg
3	SIGRAPREG® C U255-0/NF-E310/30%	0 deg
4	SIGRAPREG@ C U120-0/NF-E704/35%	90 deg
5	SIGRAPREG® C W200-TW2/2-E323/45%	45/-45 deg
Symmetry	EN AW-5052 Honeycomb	
6	SIGRAPREG® C W200-TW2/2-E323/45%	45/-45 deg
7	SIGRAPREG® C U120-0/NF-E704/35%	90 deg
8	SIGRAPREG® C U255-0/NF-E310/30%	0 deg
9	SIGRAPREG® C U255-0/NF-E310/30%	0 deg
10	SIGRAPREG® C W200-TW2/2-E323/45%	45/-45 deg





Material Data

- Make sure the colors in the TAB Chassis Pics matches that with this chart
- Make sure to correctly reference the right material values (custom/'other X') to their respective TAB's Laminate Testing and Perimeter Shear Strength tabs
 - Can be directly cross-referenced, instead of manual value input
- Most <u>front hoops</u> currently use some aluminum, we tend to look more closely to your values here!
 - If FH is made of an alternative material, is welded and/or brackets are welded to the FH, teams must provide the values of the yield and ultimate strength in welded condition

Material Spreadsheet Code	Steel	Aluminium 1	Aluminium 2	Composite 1	Other 1	Other 2	Other 3	Other 4	Other 5	Other 6
Material name	Steel	Aluminium 1	Aluminium 2	T3.5 Laminate	FBHS	FBH	SIS hor.	MHBS	SHB	FHB
Colour in Chassis Pics										
Youngs Modulus, E	2.00E+11	7.00E+10	7.00E+10	0.00E+00	0.00E+00	0.00E+00	3.00E+00	4.00E+00	5.00E+00	6.00E+00
Yield strength, Pa	3.05E+08	1.00E+00	2.00E+00	0.00E+00	0.00E+00	0.00E+00	3.00E+00	4.00E+00	5.00E+00	6.00E+00
UTS, Pa	3.65E+08	1.00E+00	2.00E+00	0.00E+00	0.00E+00	0.00E+00	3.00E+00	4.00E+00	5.00E+00	6.00E+00
Yield strength, welded, Pa	1.80E+08	1.00E+00	2.00E+00	N/A	N/A	N/A	N/A	N/A	N/A	N/A
UTS welded, Pa	3.00E+08	1.00E+00	2.00E+00	N/A	N/A	N/A	N/A	N/A	N/A	N/A
UTS shear, Pa	2.19E+08	1.00E+00	2.00E+00	0.00E+00	0.00E+00	0.00E+00	3.00E+00	4.00E+00	5.00E+00	6.00E+00

Material Data Sheet

• Steel properties based on baseline steel and are given in Rule T3.2.4 . All steels are treated equally and are to use the above properties.

• Results from extra physical test required by T3.5 should be entered in the "Other x" materials column as required.

• If you are using TAB "T3.5 Different Layups" to create different/multiple composite layups other than your SIS-Layup from TAB "T3.5 Laminate Test",

you must fill out manually the columns at "Other X" in this Material Data Sheet TAB.





- Everything/every cell that is YELLOW requires input
- Everything/every value that you actively put in the SES/IAD, needs a proof of measurement / calculation
 - Preferably directly (structured) within the available space below and in the same TAB
- Every TAB that requires to report the 'minimal or most critical section' needs proof that this is indeed the 'minimal or most critical section'
 - Include an image that illustrates this!
- The SES does (currently) not ask about insert design other than MH/MHB/FH/Accumulator/primary structures attachments
 - What about suspension? Anything that impacts your effective panel height.







23.10.23

Significant Changes

- A2.2 First year vehicle: significant changes in the chassis structure to its predecessor must be shown
 - Include images highlighting which structures are different, and tables in case of laminate changes
- Significant changes (multiple!) are e.g. material type (different lay-up), geometry (dimensions, shape, angles)
- The same mould (previous years) can be used, but new laminate structure (ply/core) is then required
- Changes to the MH and MHB do not fulfil the intent of the rule



T3.8 Main Hoop Tubing, T3.9 Front Hoop Tubing & T3.10 Main Hoop Bracing

Rather straight forward:

- Up to you to figure out which tubes are rules compliant and sufficient (T3.2 only provides a **minimum** dimension, irrespective of overall structure geometry)
- Include image with dimensions of tube

lain Hoop Structural Equivalency - note, only steel may be used

	Material Prope	erty				Baselin	ie	Your Tube	1
	Material type					Steel		Steel	1
•	Front Hoop Strube shape qu	livalency				Round		Round	1
/)	laterial name /	grade				Steel		Steel	1
	Material Property			Baselin	e 🚽	our Tubo	11	2.00E+11	
	Material type			Steel	S	iteel	8	3.05E+08	
	Tube shape			Round	F	lound	8	3.65E+08	5
Main Hoop Bra	Cinerial And Revealed Steel may	be used		Steel	5	iteel	18	1.80E+08	1
				0.001	11	2.00E+11		3.00E+08	
Material Property		Baseline	Your	Tube	08	3.05E+08			_
laterial type		Steel	Steel		08	3.65E+08		25	; <mark>]</mark>
ube shape		Round	Round		08	1.80E+08		2.5	<u>;</u>
/laterial name /grad	le	Steel	Steel		08	3.00E+08			_
oungs Modulus, E		2.00E+11	1 2.0	00E+11				Your Tube	
field strength, Pa		3.05E+08	3.0	05E+08	25	25		0.025	j -
JTS, Pa		3.65E+08	3.6	65E+08	.5	2.5		0.0025	i
rield strength, weld	ed, Pa	1.80	1.8	30E+08				1.1322E-08	5
JTS welded, Pa		3.00E+08		0E+08	Y	our Tube		2.26E+03	10
					25	0.025		176.7	10
ube OD, mm		25.4	4	25.4	25	0.0025		5.39E+04	10
Vall, mm		1.6	6	1.60	8	1.13222E-08)4	6.45E+04	10
					3	2.26E+03	100	3.18E+04	10
		Baseline	Your	Tube	.7	176.7	100.	5.30E+04	10
DD, m		0.0254	4	0.0254	04	5.39E+04	1003	1.32E+03	10
Vall, m		0.0016	6	0.0016	04	6.45E+04	1002	1.22E-02	10
m^4		8.51E-09	8.	51E-09	04	3.18E+04	1000	8.04E+00	10
-		1.70E+03	3 1.7	70E+03	100	5.30E+04	100.	0	
Area, mm ²		119.6	6	119.6	100	1.32E+03	100.	0	
ield tensile strengt	h, N	3.65E+04	4 3.6	65E+04	100	1.22E-02	100.	0	
JTS, N		4.37E+04	4.3	37E+04	100.0	8.04E+00	100.	0	
rield tensile strengt	h, N as welded	2.15E+04	1 2.1	15E+04	100.0				
JTS, N as welded		3.59E+04	1 3.5	59E+04	100.0				
Max load at mid spa	an to give UTS for 1m long tube, N	9.78E+02	2 9.7	78E+02	100.0				
Max deflection at ba	aseline load for 1m long tube, m	1.20E-02	2 1.	20E-02	100.0				
nergy absorbed up	o to UTS, J	5.86E+00	5.8	36E+00	100.0				

T3.2 Minimum Material Requirements

T 3.2.1 Table 4 shows the minimum requirements for the members of the primary structure if made from steel tubing.

Item or application	Minimum wall thickness	Minimum cross sectional area	Minimum area moment of inertia
Main and front hoops, shoulder harness mounting bar	2.0 mm	173 mm ²	11320 mm ⁴
Impact structures, front bulkhead, roll hoop bracing, driver's restraint harness attachment (except as noted above)	1.2 mm	119 mm ²	8509 mm ⁴
Front bulkhead support, main hoop bracing supports	1.2 mm	91 mm ²	6695 mm ⁴

Table 4: Minimum Material Requirements

T3.9 Front Hoop Tubing

- Front hoop: make sure the material data is correct, based on datasheets and/or physical test
- If you weld a FH:
 - you should apply a heat treatment, if it can be done **professionally** (certificate required)
 - If not, physical testing required, in 'normal' and 'as welded' condition
 - Explain your design!

Front Hoop Structural Equivalency

Material Property	Baseline	Your Tube
Material type	Steel	Aluminium 1
Tube shape	Round	Square
Material name /grade	Steel	Aluminium 1
Youngs Modulus, E	2.00E+11	7.00E+1
Yield strength, Pa	3.05E+08	2.40E+08
UTS, Pa	3.65E+08	2.75E+08
Yield strength, welded, Pa	1.80E+08	2.40E+08
UTS welded, Pa	3.00E+08	2.75E+0

wall, mm	2.4	2	
	Baseline	Your Tube	1
OD, m	0.0254	0.0314	1
Wall, m	0.0024	0.002	
l, m^4	1.1593E-08	3.404E-08	
El	2.32E+03	2.38E+03	102.
Area, mm ²	173.4	235.2	NA
Yield tensile strength, N	5.29E+04	5.64E+04	106.
UTS, N	6.33E+04	6.47E+04	102
Yield tensile strength, N as welded	3.12E+04	5.64E+04	180.
UTS, N as welded	5.20E+04	6.47E+04	124
Max load at mid span to give UTS for 1m long tube, N	1.33E+03	2.38E+03	178
Max deflection at baseline load for 1m long tube, m	1.20E-02	1.17E-02	97.
Energy absorbed up to UTS, J	7.98E+00	2.49E+01	311.

BACK to COVER SHEET

Tube OD, mm

The FRH is made from a machined "U" section and a flat cover as shown on the drawing. The two parts are welded to each other. Material is aluminium EN AW-5082, which has a "very good" weldability according to the provided data sheet. After welding, the FRH was heat treated (T6) to restore the material properties lost by the welding process. Documentation of the heat treatment is attached. The welds have not been grinded down or otherwise mechanically altered as specified in T 3.2.7, the surface of the whole FRH has been sand blasted though in order to improve the bonding.





Kunde:	AM2 R	acing Tea	II, ETH	2							
Wärmebelt	andlun	E		Vera	üten	w	armas	slagern	-		
			50		ist	50	8	ist.	5	Ы	ist
Aufheizges	chw. 'C	/h							-		
Behandlun Holteroit h	gstemp	°C	53	0	530	17	0	170	-		-
Abkühlung	spesche	r. 'C/h		-	0	- "	-	- 10	-		-
Abschreck	en Medi	ium	Was	uer .	Wasser						
Sollwerte r	hach der	r Behandi	ung	Ober	fláchenháro	r=n.a.					
Ernst	Hand	Esatest		027	Härtew	erte	Mobi	iler Härteg	nüfer	Bas	Ael
Bestell-	Pos.	Ancahl	1		Stickber	eichnu			Mater	4	Ist-Hirte
nummer	1	1	A231	11-12	FRH				3.2315	-	



T3.10.5 T3.5 MHoop Brace Spt



- New rule T3.5.4 not quasi-isotropic laminates, oriented accordingly
- We look for: clear measurements, subtracting holes, are larger hole edges laminated, is (in top view) the MHB attached to the laminate on a stiff edge/corner?
- In case of proof of equivalence: is the **structure geometrically** stiff, load path reasonable? shots / Images proving all panel dimensions and any addition
- Guidance notes!







Length 158.9143 mm 🖷 🔞

General: How-to El calculation

- Based on T3.4.1 you may calculate the actual flexural rigidity (EI) of the structure for some areas in the chassis within your SES
- But it is not intended to calculate the EI of the entire cross-sectional area incl. core, but from the skin layers!
- Indication for wrong calculation (besides 10x higher EI, than flat panel calculation) is also the big section area used for your UTS calculation



T3.11 T3.5 FHoop Bracing & T3.14 T3.5 FBH Spt. Structure



- We look for:
 - proper loadpath, not only in side view, but also in top view, triangulation
 - FHBS with 'damper cut-out' with depth (z-direction) > 50 mm should have at least an angle of 30 degrees to the connection areas
 - 'Damper cut-out' is always reviewed more closely, prevent another review by already performing additional calculations in advance
 - Alternative calculation allowed!



T3.13. T3.5 Ft Bulkhead

- If the core thickness is reduced towards the cutout edges, the FBH structure has no uniform stiffness
 - Effective 'cut-out' dimensions must meet the area where the actually installed laminate has the same structure as the tested laminate
 - Take into account manufacturing difficulties (e.g. core indentation)
- T1.1.5 in front view, the front bulkhead (together with the AIP) must cover the driver's feet
- Problems often only noticed at the event's technical inspection
- IN 5.1.1: bring the representative bulkhead/test fixture of the IA test
 - Common issue: not the same attachments as reported!



T3.15 T3.5 SIS

- Monocoque
 - Equivalent proof using the Flexural Rigidity NOT allowed in the SIS! (T3.4.1)
 - Account for holes (TS, cooling, draining)
 - Consider the whole structure:
 - Chassis is as strong as its weakest point
- Tubular frames:
 - Ensure proper connection to hoops, e.g. T3.14.1
 - Check ride height play in combination with upper member!
 - Common error: upper SIS too high with proper ride height (>30mm)





T4.5 Shoulder Harness Bar



- Last year more detailed calculations required, despite minimum requirements (T3.2):
 - These requirements do not consider overall geometry of the structure itself
 - Problem: the minimum requirements are not sufficient for a typical shoulder harness span width
 - Required solutions: larger OD tube, additional supports closer to load application (reduce SHB length)
- Assume: load applications >50 mm away from a node require additional calculations
- For harness attachments, there are <u>additional</u> requirements related to the chassis design (T4.5, T5.5.3)
- Dont 'optimize' by making the car smaller, the driver should still fit!
- Guidance notes!



Beam stress scales with the area moment of inertia, increasing OD has more impact on stiffness than increasing wall thickness (decreasing ID)

$$I = \frac{\pi (D^4 - d^4)}{64}$$

Item or application Main and front hoops, shoulder harness mounting	Minimum wall thickness 2.0 mm	Minimum cross sectional area Minimum area moment of inertia 173 mm ² 11320 mm ⁴		
Outer diameter [mm]	Wall thickness [mm]	Inertia [mm^4]		
25	2,5	11320		
50	2,5	105507 (factor 9)		
25	5	16689 (factor 1,5)		

23.10.23

27

EV5.5.1, EV5.5.2, EV4.4 ACPS TSPS Side/Rear





T3.5 Laminate Test – and all following











- Material should be the baseline material (T3.2.4)
- Material dimensions should be the same as mentioned in T3.2.1 for the related structure
- For SIS: two baseline material tubes must be tested
- Energy absorbed for SIS must be at least 65 J
- Only alternative size for laminate panel allowed in TSAC: TABs EV5.5.4 Alt. Matl – 3pt Bending
- Proper documentation
 - Marked on laminate
 - Testing rig
 - All information the graph (which values used etc.)

T3.16 MH & MH B'ing, FH & FH B'ing Attachments

- Bad design:
 - Allow local loading/deformation of the tube instead of loading entire tube section
 - Stress concentration on end points of the weld
 - Poor gusseting
- Good design:
 - Good load distribution to hoop, all around
 - Stress concentration captured by surrounding bracket & weld structure
 - Bracket is stiff at monocoque attachment point



Local reinforcement (creative bookkeeping), last-minute solution. Ensure welding quality



Localised loadpath into hoop, weld attachment not stiff. All welds on single line, poor bending/torque resistance



attachment not stiff, minimal gusseting





T3.9.6 Laminated front hoop

- Consider **all three directions** load cases
- Ensure that production matches design
 - Document lamination (will be asked during review / inspection)
 - Are the dimensions matching?
- Ensure proper production
 - If made incorrectly, a laminated front hoop is a weakpoint instead of a strong point!
 - Ensure proper bonding to both monocoque skins
 - Proper finish on core material & tight fit















Author: Formula Student Germany

T3.16 & T3.17.5 Primary Structure Attachments

- Formula will be adjusted
- Guidance note will be created



	OTHER Panel/Plate Attachment	FAIL			
	Perimeter, mm	1400			
	Required No. of fasteners	7			
	Fastener dia., mm	8,0 PASS			
A N/A	No. of fasteners	7 PASS			
IL N/A	Max. Distance between bolt centres	29 FAIL			
1800					
9	Charle distribution of bolts and also	have to be low			
	Check distribution of bolts and ela	iborate below.			
5.0 CHECK					
10 CHECK	Material 1				
entres 242 FAIL	Skin / Material type	Composite 1			
	Skin / Material name	T3.5_Laminate			
and olaborate below	Washer/backing plate perimeter, mm	200			
and elaborate below.	Washer/backing plate thickness, mm	2 PASS			
	Insert Perimeter, mm	200			
	Edge distance, mm	60			
	Skin thickness, mm	2			
	Skin thickness, mm	2			
1,5	Skin shear strength, MPa	50			
	Material 2	Material 2			
2	Skin / Material type	Composite 1			
2	Skin / Material name	T3.5_Laminate			
FAIL	Washer/backing plate perimeter, mm	200			
	Washer/backing plate thickness, mm	2 PASS			
	Insert Perimeter, mm	200			
0	Edge distance, mm	60			
0 FAIL	Skin thickness, mm	2			
0 FAIL	Skin thickness, mm	2			
0 FAIL	Skin shear strength, MPa	50			
hment point proving					
ed above:	Perimeter shear strength, kN	40 PASS			



AIP to Bulkhead Attachmer

Max. Distance between bolt (

Check distribution of bolts

Perimeter, mm Required No. of fasteners

Fastener dia., mm No. of fasteners

AIP Thickness, mm Washer/bolt perimeter, mm Skin thickness, mm Insert Perimeter, mm Skin thickness, mm Backing plate thickness, mm Backing plate perimeter, mm Edge distance, mm Skin shear strength, MPa Perimeter shear strength, kN Perimeter shear strength, kN Tearout strength, kN Insert images of each attact the remaining values enter

T4.5 Harness Attachments

Common issues:

- Not at stiff node, no support
- Lowest chassis member: check for ride height
- Calculations required for bracket as well (not just the beam to which it is attached, for monocoques physical test required)



EV5 Accumulator Container, Acc. Stack Construction

- See Session Accumulator 16:00-17:00, after 'Brezel Break'
- EV5.5 Tractive System Energy Storage Mechanical Configuration
 - Several changes in the rules this year, please read them carefully, as they all are to be documented in the SES!
- Accumulator designs are tricky do document & check in SES
 - Make sure your design is clearly structured
 - Make sure the produced TSAC is the same as what was presented in the SES
 - Improper manufacturing or deviation from the design can be showstoppers! TSAC's are difficult to adjust on site

EV5.5. Acc. Attachments

- Complicated designs require complicated equivalency!
 - Keep it simple!
 - Hand calculations are preferred (but can be accompanied by FEM)
 - Make them as designed in SES



Monocoque Lap Joints

Monocoque Lap Joints

Joint Status	PASS	
Single piece monocoque?	No	
Test sample lap area, mm^2	660.0	
Lap joint shear strength, MPa	22.28	
Overlap length "w", mm	22	
Load/unit length, N/mm	490.2	
Skin UTS, MPa (from 3pt bend)	234	
Skin thickness, mm	0.6	
Load/unit length	133	
Safety Factor	3.67	









- Pictures of all measurements (lap area, width and height)
- Pictures of the lap joint in the monocoque (when available, otherwise a sketch)
- Force-displacement curve

Welded Tube Inserts

Welded Tube Insert Equivalency - required for all mandated tubes per T3.7.6 with drilled holes > 4mm dia.

If you have more than one type of insert please copy this tab

	1		1
		Your Tube	
Material Property	Your Tube	+ Insert	
Material type	Steel	Steel	
Tube Shape	Round	N/A	
Material name /grade	Steel	Steel	
Youngs Modulus, E	2.00E+11	2.00E+11	
Yield strength, Pa	3.05E+08	3.05E+08	
UTS, Pa	3.65E+08	3.65E+08	
Yield strength, welded, Pa	N/A	1.80E+08	
UTS welded, Pa	N/A	3.00E+08	
		•	
Tube OD, mm	25		
Wall, mm	1.5		
		Your Tube	
	Your Tube	+ Insert	
OD, m	0.025		
Wall, m	0.0015		
l, m^4	7.677E-09	1.03E-08	
El	1.54E+03	2.06E+03	134.4
Tube area, mm^2	110.7	150.0	NA
Insert area, mm^2	NA	58.0	NA
Yield tensile strength, N	3.38E+04	5.62E+04	166.4
UTS, N	4.04E+04	7.22E+04	178.5
Max load at mid span to give UTS for 1m long tube, N	8.97E+02	1.21E+03	134.4
Max deflection at baseline load for 1m long tube, m	1.22E-02	9.05E-03	74.4
Energy absorbed up to LITS	545E+00	7.33E+00	134.4

Note - the calculations above take account of the "as welded condition" strength reduction Insert calculations or CAD screenshot proving tube + insert area and second moment of area below.



Steering Rack Collars







Successful IAD submission

FSG Academy – Main Workshop for FSG 2024 on 21st of October 2023 at Schaeffler in Herzogenaurach

Agenda

From 'front to back':

- IA
 - Standard / 'Own' Design
- IA to AIP Attachment
- AIP
- AIP to Front Bulkhead Attachment
- (Dynamic) Testing





This form must be completed and uploaded to the "My Team" area on the FSG website no later than the date specified in the Action Deadlines. A printed copy of this form must be presented together with the ve

The Impact Attenuator Data (IAD) and supporting calculations must be submitted electronically in Adobe Acrobat format (*.pdf)



Attach Proof of Impact Attenuator

If the IA (Impact Attenuator) is a "Standard IA Design", the following points must be included:

- 1. The first page must always be this FSG Impact Attenuator Data Form
- 2. The report must be written in "engineering style" (e.g. contents, captions, symbols and abbreviations, page
- 3. Design of IA, positioning on the AIP and IA volume (T3.17.2) above the ground (dimensions in mm) 4. Method for attachment of the IA to the AIP (including data sheets e.g. if it bonded together)
- 5. Dimensions of the front bulkhead (dimensions in mm)
- 6. Proof of additional diagonal or X-bracing in the bulkhead or equivalent per T3.17.7, if applicable
- 7. Design of the AIP (material, thickness and dimension in mm)
- 8. Method for attachment of the IA assembly (AIP) to the front bulkhead 9. Receipt of the material, a packing slip or letter of donation of the IA
- 10. Pictures (or sketches) of the attachment on the car
- 11. Please comply with the particular rules for front wings and positioning of non-crushable such as sensors, if applicable

FORMULA STUDENT GERMANY Impact Attenuator Data Form - Teams's Own IA Design 101000 000 0 0 0 0 000 0 0 0 0 0



This form must be completed and unloaded to the "My Team" area on the FSG website no later than the date specified in the Action Deadlines. A printed copy of this form must be presented together with the vehicle a

The Impact Attenuator Data (IAD) and supporting calculations must be submitted electronically in Adobe Acroba format (".pdf)



Attach Proof of Impact Attenuato

If the IA (Impact Attenuator) is a "Team's Own IA Design", the following points must be included

- 1. The first page must always be this FSG Impact Attenuator Data Form
- 2. The report must be written in "engineering style" (e.g. contents, captions, symbols and abbreviations, page numbers, experimental setup, evaluation

 FS Germany accepts only dynamic impact attenuator tests (e.g. sledge test or drop down) with real test data (shown in rule T3.19.1), including impact attenuator, anti intrusion plate (AIP), front bulkhead and optionally (per T3.19.4) the front wing, other non-crushable object(s) and/or structurally representative dummies thereof in front of the AIP.

- 4. Design of IA, positioning on the AIP and IA volume (T3.17.2) above the ground (dimensions in mm)
- 5. Method for attachment of the IA to the AIP (including data sheets e.g. if it bonded together)
- 6. Dimensions of the front bulkhead (dimensions in mm)
- 7. Design of the AIP (material, thickness and dimension in mm) 8. Method for attachment of the IA assembly (AIP) to the front bulkhead
- 9. Description of the test set up (including sensor, data acquisition system, test fixture)
- 10. If alternative materials are used for the AIP, equivalency to T3.17.3 must be proven by physical testing as in T3.19.2. Test fixture must be made from the same materials as the intended chassis (consistent with SES)
- 11. If the test is accomplished at a company or research center, a letter of conformity must be attached.
- 12. If the test is accomplished at the university, an official of the university (with contact details) must sign a etter of conformity (must be attached to the report)
- 13. Table of measured results of the dynamic impact attenuator test: test speed, absorbed energy, graph of average deceleration and peak deceleration over an interval of time, permanent deflection of the AIP
- 14. Receipt of the material, a packing slip or letter of donation of the IA
- 15. Pictures before / after the dynamic impact attenuator test

2023-01-26 | iad@formulastudent.de | Rev-cda0855



Impact Attenuator (1/2)

General

- recap of positioning of IA: 'for any portion of the required 100×200 ×200 mm³ volume' < 350 mm (T3.17.3)
- 'in XY-plane'
- closed front section
- standard or own design? → effort vs. gain
- Standard Design:
 - 200 x 100 x 200 mm
 - pre-crushed side to AIP
- 'Own' Design:
 - alternative AIP designs must have equivalence to T3.17.3 and be demonstrated by physical testing as in T3.19.2 'crash test'









Impact Attenuator (2/2)

- Standard design:
 - Honeycomb



Foam



• Own design:



Attachment of IA to AIP (1/2)

- 4x 8mm grade 8.8 bolts **OR** adhesive attachment >60 kN in any direction (T3.17.5)LOCTITE
 - Note: due to T3.2.8, effectively 120 kN!
 - For standard designs: T3.17.7 adhesive with min. shear strength 24 MPa (calculation still required)
 - Examples: Loctite EA9466, Loctite EA E 20HP,
 - 3M Scotch-Weld DP490, 3M Scotch-Weld EC9323 B/A
 - and many more...



Always use engineering judgement, e.g. is this adhesive suitable for CRFP's/aluminium etc.?



Advice: DP490 very good in combination with CFRP's



TYPICAL PERFORMANCE OF CURED MATERIAL **Adhesive Properties** Cured for 5 days @ 22 °C Lap Shear Strength, ISO 4587: Steel (grit blasted) Aluminum (abraded) (psi) Aluminum (anodised) (nei) alvanized Steel (Hot Dipp Stainless steel Polycarbonate N/mm² 1.6 Nylon (psi) (230) N/mm² 11.3 Wood (Fir) (1.635) (psi)

Cured for 7 days @ 22 °C, 1.2 mm thick film

Shore Hardness, ISO 868, Durometer D

Tensile StrengthASTM D 882

Physical Properties



Technical Data Sheet

LOCTITE EA 9466™

Exposure Time, hours

Known as LOCTITE[®] Hysol[®] 9466™ August 2019





Attachment of IA to AIP (2/2)





Insert in FBH





AIP arranged diagonally



Bad welding



Bad welding

complex design of attachment points



sloping AIP



IA glued directly to FBH



Author: Formula Student Germany

AIP

Standard:

- 1.5 mm steel/4.0 mm aluminium
- Show drawing with dimensions, thickness and material and distances of holes for screws



'Own' Design:

- show materials, lay-up, joining methods, manufacturing method
- present datasheets and receipts





AIP to Front Bulkhead Attachment

- General: see T3.16.6
- Standard:
 - Welded 'in all directions', continuous welding
 - Bolted







- 'Own' Design:
 Special designs
 - No holes in FBH
 - IA directly glued to FBH



 T3.19.3
 Structural nose (only quasi-static test required)





(Dynamic) Testing (1/4)



- T3.19.1 'Equivalent (higher) test velocities are only allowed to accommodate for a lower total testing mass, as long as the energy absorbed is 7350 J or more.'
 - If these requirements cannot be met, a team must use the standard IA.
 - Higher total testing mass than 300 kg are not allowed.
- T3.19.5 only dynamic testing









(Dynamic) Testing (2/4)

T3.19.4 - Crushable objects in front view of AIP require physical testing



front wing and mountings

sensors and mountings



(Dynamic) Testing (3/4)

- Please show the following diagrams in the IAD:
 - average deceleration and peak deceleration over time absorbed energy over time (required)
 - absorbed energy over time
 - force over displacement or force over time











Author: Formula Student Germany

(Dynamic) Testing (4/4)



- IN 5.1.1: 'IA test piece and (representative) test fixture (except for teams with 'standard' IA) must be presented at inspection
 - Possible rules change in future (AIP & FBH must remain intact) analogue to FSAE







General Directions for IA Documentation

- Proper documentation includes:
 - Values backed up with equations, units
 - Pictures displaying different views (ISO, side, top...) before AND after test
 - Pages, tables, figures numbered and referenced
 - Datasheets and relevant values highlighted
- Review is much quicker if we do not need to 'search'



Successful SESA submission

FSG Academy – Main Workshop for FSG 2024 on 21st of October 2023 at Schaeffler in Herzogenaurach

Targets of SESA

Some repeating required:

Targets of "SESA"



- Your monocoque is a product of your team and your team is responsible for this product
- The required SESA is a quality check by an external third party reviewer if your product meets the given requirements stated in the rules for FS Germany
- The external third party reviewer should have a qualification and knowledge about laminate structures (A5.7.2)
- Reviewing an SES for all given points in the SESA requires at least 3-4 hours which the external third party reviewer needs to invest.

Targets of "SESA"



- The external third party reviewer should check all the given information in each TAB of your SES for all given points in the SESA
- If the external third party reviewer find some issues he should use the change log of the SESA. Then the team should fix the issues and the external third party reviewer should recheck these points again before submitting the SES to FS Germany.
- The complete changelog of your SESA process must be provided with the SESA

Personal note:

Just putting the check marks in the SESA without proper control is in violation of the FSG rules and will lead to a de-registration of the competition. FS Germany can/will require a new external third party reviewer.

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2019-11-16

The SESA is also your chance as a team to ask and get valuable feedback from a professional.

2019-11-16

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Thank you for your attention

